



# Laser Ranging to Measure LightSail 2 Orbit Raising

David Spencer, Purdue University  
Bruce Betts, The Planetary Society

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# LightSail 2 Mission Overview

- The LightSail 2 project will demonstrate controlled solar sailing in Earth orbit from a 3U CubeSat platform
  - First mission to change orbital energy via solar sailing
- Scheduled for launch no earlier than Nov. 30, 2018 as part of the Space Test Program-2 payload on the SpaceX Falcon Heavy launch vehicle
  - Injected into a 720 km, 24 deg inclination orbit
  - Deployed by Prox-1 microsatellite 7 days after launch
- Through controlling sail orientation relative to the sun, the orbit apogee will be raised  $\sim 600$  m/day for one month following sail deployment





# LightSail 2 CONOPS



LS2 Deployment from Prox-1  
LS2 Initial Acquisition & Checkout  
(L+7 days)



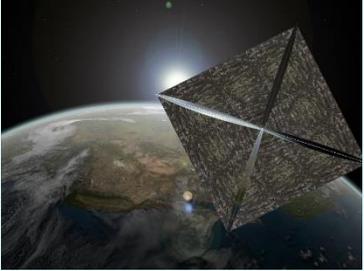
Solar Panel Deployment  
(L+12 days)



Solar Sail Deployment  
(L+13 days, Duration 2.5 min)



Sail Control: Apogee Raise  
(L+13 – L+43 days)



# Flight Ops Sequence of Events

Days Past Launch	Event	Notes
L+7	P-POD deployment	Prox-1 timer is set to deploy LS2 7 days after launch vehicle injection. The 7 days will allow separation of the 11 spacecraft deployed into the 720 km orbit so that they can be individually identified.
L+7 – L+12	On-orbit checkout and orbit determination	Evaluation of LS2 engineering subsystems and cameras. Orbit determined by the 18 <sup>th</sup> Space Control Squadron located at Vandenberg Air Force Base, California (TLEs) and the International Laser Ranging Service.
L+12	Deploy solar panels	Panel deployment will be commanded following completion of on-orbit checkout
L+13	Deploy solar sail	Sail deployment will be commanded after verification that EPS performance with panels deployed is as expected. Sail deployment takes about 2.5 minutes. Downlink images.





# Flight Ops Sequence of Events (Cont.)

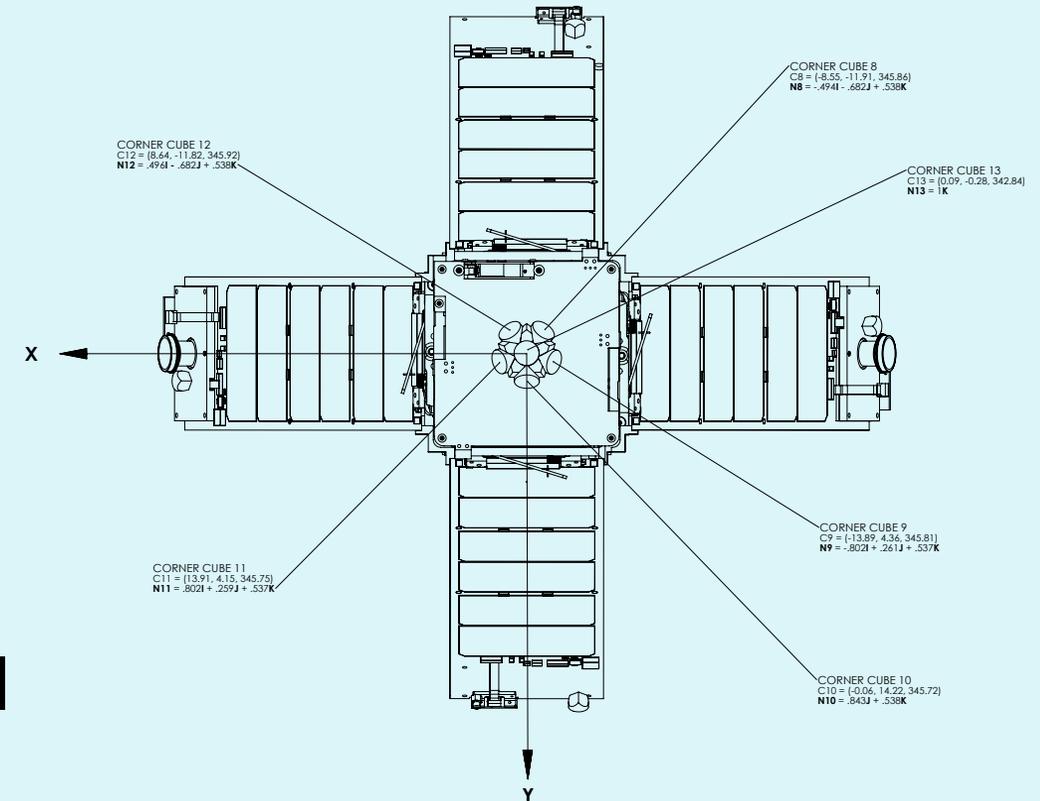
Days Past Launch	Event	Notes
L+13 – L+43	Controlled solar sailing	90 deg sail rotations every half orbit, controlled via momentum wheel and torque rods. Apogee raise of ~700 m per day expected.
L+43 – L+57	Extended mission phase	Possible evaluation of alternate solar sail control strategies
L+57 – L+180	Deorbit phase	Continued downlink of stored images and telemetry. Possible additional imaging, and characterization of uncontrolled sail orientation.
> L+180	Reentry and End-of-Mission	Timing of reentry is very uncertain. Reentry could occur from 6 months to 3 years past launch.

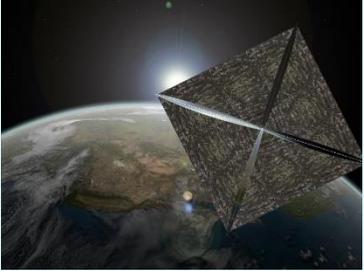




# LightSail 2 Laser Ranging Corner Cube Placement

- LS2 includes 13 corner cubes for laser ranging
  - One mounted on each of the deployed solar panels (4 total)
  - One mounted on each of the +X, -X, -Y faces (3 total)
  - Six mounted as an array on the +Z face
- Only +Z array visible prior to panel deployment





# Laser Ranging Plan

- Laser ranging is planned for precise orbit determination both before and after sail deployment
- Only stations within  $\pm 30^\circ$  of equator can observe LS2
- Will request TLE updates twice per day once sail is deployed
- Laser ranging will be challenging, due to the changing orbit
  - Attempts to track LightSail 1 (summer 2015) were unsuccessful
  - Corner cube array on spacecraft +Z axis added for LightSail 2
- The LightSail 2 team greatly appreciates the efforts of the laser ranging community in helping the project to determine solar sailing performance.

